



MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

MRB FEC							TASK TYPE	ORIGIN
SKILL	B1					TOTAL	ETOPS / CHANGE INDIVIDUAL GROUP NUMBER	MOD
M/H	600.00					600.00		
PEOPLE	1					1		

REF. INTERVAL	ZONE	NRWI TENDENCY
Hours: 0 Cycles: 0 Days: 0		

RELATED TASK CARDS

BEFORE	N/A
AFTER	N/A

PLANNING REQUIREMENTS FOR THE TASK CARD ACCOMPLISHMENT

No planning requirement assigned to this task card

REFERENCES

Reference Number	Revision No	Revision Date
AD 2007-25-09	00	14.01.2008
ASB A310 53-2125	01	
EASA 2007-0111	00	09.05.2007

MATERIALS

PN	DESCRIPTION	Qty	
343A100KN2	HYDRAULIC LOAD CELL	2	SPARE
343A160KN2	LOAD CELL	1	SPARE
343A400KN2	LOAD CELL	2	SPARE
343A63KN2	HYDRAULIC LOAD CELL	1	SPARE
532125D01R02	KIT	1	SPARE
98A07003903001	FR 17 R/H FITTING	2	SPARE
98A07003906200	ADAPTOR	6	SPARE
98A07003907200	ADAPTOR	5	SPARE
98A07003909200	ADAPTER	2	SPARE
98A07003910200TK	ADAPTOR	1	SPARE
98A07004002000	JACKING ADAPTER - JACKING	2	SPARE
CBSR6-3N1	REAMER	2	SPARE

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	1/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20





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PN	DESCRIPTION	Qty	
MS02-2	AUXILLARY JACK	2	SPARE
MS02-3	JACK	2	SPARE
RTA075009LHTK	CRADLE	1	SPARE
RTA075010RHTK	CRADLE	1	SPARE

Note: Non-specific/standard materials are included here for information only and must be used in line with the requirements given in the task procedure.

TOOLS

No tool assigned to this task card

CHECKED BY

NAME, SURNAME:	Sign:	Date:	STAMP:
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ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	2/32	

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Perf.By Insp.By

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

ACN	MSN	IPC	Work Order Number	Rev No.	Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	3/32	

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TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58**THY TEKNİK / TURKISH TECHNIC**

Approval References : EASA.145.0276 SHY-145/JAR-145: TR-00001; FAA: TQKY144F


TEKNİK YETKİ EMRİ
 ENGINEERING ORDER

 Sayfa No
 Page No

1 / 27

TYE No E/O No	1-5316-002	Orj. TYE Tarih Org. E/O Date	29.12.2008	İşlem Türü Work Type	MODIFICATION
Son Revizyon No Last Revision No	01	Son Rev. Tarihi Last Revision Date	29.06.2010	Müşteri Customer	AN
Konu Description	FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58				
Sebebi Reason	AS A RESULT OF A WIDE SPREAD FATIGUE DAMAGE (WFD) CALCULATION ON A310 AIRCRAFT, IT WAS FOUND THAT A MODIFICATION OF THE UPPER FUSELAGE CIRCUMFERENTIAL JOINT AT FR55/58 IS REQUIRED TO ENABLE THE AIRCRAFT TO REACH THE EXTENDED SERVICE GOAL (ESG).				
Referanslar References	1.) EASA 2007-0111 00, AD 2007-25-09 00 2.) BA310 53-2125 01				
İlgili Atelyeler Relevant Shops	A/C MAINTENANCE SHOP, REPAIR MAINTENANCE SHOP, CABIN INT. SHOP			MKF No Ecos No	53-0012
Uygulama Tarihi Embodiment Date	BEFORE 33000 FC OR 93500 FH, WHICHEVER OCCURS FIRST.				
Kapsam Effectivity	CAQ CAV				
Sonuç Bildirim Formu / Result Information Form		☒ İsteniyor/ Requested		☐ İstenmiyor/ Not Requested	

İşçilik-Uçak (Adam-Saat) Labor-A/C Application (Man-Hour)	600	Birim TYE Uygulama Süresi EO Elapsed Time (Hour)	240
İşçilik-Atl. (Adam-Saat) Labor-Shop Application (Man-Hour)	0	Ağırlık Değişimi Weight Change (lb)	20.5
Birim TYE Malzeme Maliyeti Unit E/O Material Cost (\$)	3216	Moment Değişimi Moment Change (lb.inch)	2123.39
Toplam Malzeme Maliyeti Total Material Cost (\$)	6432	Uyg. Uçak/ Komp./ Teçhizat Sayısı Qty. Of Aircraft/ Comp./Equip. To Be Modified	2

Notlar/ Notes		Notlar/ Notes	
Planlama Bilgisi Planning Info	☒ 01,02,03	Uçuş İşletme/Uçuş Eğitim Bilgisi Operational Info	
MPDF Material Procurement	☒	Teknik Eğitim Bilgisi Technical Training Info	
Warranty Warranty		Elektrik Yük Değişimi Electrical Load Change	
Revizyonu Tekrar Uygulama Retroactive Application		Uçağın Sig. Değ. Değiştiren İşl. Change in Aircraft Insurance Value	
Uçaktan Komponent Söküm/Tak. Component Replacement on Aircraft		Kritik İtem Critical Item	
Simülatör (Kokpit/Kabin) Bilgisi Simulator (Cockpit/Cabin) Info		RII Required Inspection Item	☒
İç Envantere Effect on Inventory			

Hazırlayan/ Prepared By	Kontrol Eden/ Checked By	Onaylayan/ Approved By
ABDULLAH ERTEN STRUCTURAL ENGINEER	TURGAY HAYRULLAH STRUCTURAL LEAD ENGINEER	MESUT YILMAZ STRUC AND CABIN. INT. ENG. MGR.

Uçak Tes. No A/c Reg. No	THY P/K THY Part Code	Bakım No Maint. Check No	Mühür Stamp
İş Emri No Work Order No	THY S/N THY Serial No	Bakım Tarihi Maint. Date	

Issue Date : 12.03.2008

Form No: MPM 1.8A/01A R4

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
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TASK CARD

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Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	2
Sıra No Item No	İş Tarifi Accomplishment Instructions			Uygulayan Perform.By	Kontrol Eden Inspected.By
NOTES 1: To accomplish this EO, aircraft should be on jacks. Jacking procedure is given in Para C. Procedures 1. JACKING. 2: To accomplish this EO, de-fueling procedure should be done. And also potable water tanks should be empty. For other details see Para C. 1. B) AIRCRAFT CONFIGURATION. 3: Kit price is 4640 USD. Airbus will provide Kits with 35 % reduction for purchase orders before 31 Dec. 2018. Price with reduction is 3016 USD. 4: With Rev.01 of this EO, Total KIT quantity in MPD page was changed,					
Uçak Tes. No A/C Reg. No		THY P/K THY Part Number		Bakım No Maint. Check No	
İş Emri No Work Order No		THY S/N THY Serial Number		Bakım Tarihi Maint. Date	

Issue Date : 23.05.2006

Form No: MPM 1.8A/01B

ACN	MSN	IPC	Work Order Number	Rev No.	Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	5/32	



TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	3
Sıra No Item No	İş Tarifi Accomplishment Instructions			Uygulayan Perform By	Kontrol Eden Inspected By
A.	GENERAL WARNING: MAKE SURE THAT YOU OBEY ALL THE WARNINGS AND ALL THE ACTIONS INCLUDED IN THE REFERENCED PROCEDURES. CAUTION: ALWAYS OBEY THE PRECAUTIONS THAT FOLLOW TO KEEP ELECTRICAL WIRING IN A SATISFACTORY CONDITION (ELECTRICALLY AND MECHANICALLY SERVICEABLE). WHEN YOU DO MAINTENANCE WORK, REPAIRS OR MODIFICATIONS, ALWAYS KEEP ELECTRICAL WIRING, COMPONENTS AND THE WORK AREA AS CLEAN AS POSSIBLE. TO DO THIS : • PUT PROTECTION, SUCH AS PLASTIC SHEETING, CLOTHS, ETC; AS NECESSARY ON WIRING AND COMPONENTS • REGULARLY REMOVE ALL SHAVINGS, UNWANTED MATERIAL AND OTHER CONTAMINATION. THESE PRECAUTIONS WILL DECREASE THE RISK OF CONTAMINATION AND DAMAGE TO THE ELECTRICAL WIRING INSTALLATION. IF THERE IS CONTAMINATION REFER TO ESPM 20-55-00.				
B.	PREPARATION Passenger Aircrafts - Make sure that the aircraft is electrically grounded. (Refer to AMM 24-00-00 Page block 301). - Remove the sidewall Overhead Stowage Compartments (OHSC) as necessary to get access to the work area (Refer to AMM 25-24-11 Page block 401). - Remove the center OHSC as necessary to get access to the work area. (Refer to AMM 25-24-21 Page block 401). - Remove the ceiling panels as necessary to get access to the work area. (Refer to AMM 25-23-11 Page block 401). - Remove the air conditioning duct(s) as necessary to get access to the work area. - Remove the heat and sound insulation as necessary to get access to the work area. (Refer to AMM 25-22-11 Page block 401). - Do the jacking procedure (Refer to Para C. 1.).				
Uçak Tes. No A/C Reg. No		THY P/K THY Part Number		Bakım No Maint. Check No	
İş Emri No Work Order No		THY S/N THY Serial Number		Bakım Tarihi Maint. Date	

Issue Date : 23.05.2006

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FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	4																																																																																
Sıra No Item No	İş Tarifi Accomplishment Instructions			Uygulayan Perform. By	Kontrol Eden Inspected By																																																																																
C.	PROCEDURES 1. JACKING Jacking procedure for the installation of the external doubler on the fuselage skin at FR55/58, between STGR13 LH and STGR13 RH. A) FIXTURES, TOOLS TEST AND SUPPORT EQUIPMENT <table border="1"> <thead> <tr> <th>ITEM</th> <th>PART NUMBER</th> <th>QTY</th> <th>DESCRIPTION</th> </tr> </thead> <tbody> <tr><td></td><td>343A160KN2</td><td>1</td><td>LOAD CELL</td></tr> <tr><td></td><td>343A400KN2</td><td>2</td><td>LOAD CELL</td></tr> <tr><td></td><td>343A100KN2</td><td>4</td><td>LOAD CELL</td></tr> <tr><td></td><td>343A63KN2</td><td>1</td><td>LOAD CELL</td></tr> <tr><td></td><td>98A07003910200</td><td>1</td><td>ADAPTOR</td></tr> <tr><td></td><td>98A07003906200</td><td>1</td><td>ADAPTOR</td></tr> <tr><td></td><td>98A07004002000</td><td>2</td><td>ADAPTOR</td></tr> <tr><td></td><td>98A07003909200</td><td>2</td><td>ADAPTOR</td></tr> <tr><td></td><td>98A07003911200</td><td>2</td><td>ADAPTOR</td></tr> <tr><td></td><td>98A07003907200</td><td>5</td><td>ADAPTOR</td></tr> <tr><td></td><td>98A07003906200</td><td>5</td><td>ADAPTOR</td></tr> <tr><td></td><td>D30-11</td><td>1</td><td>JACK</td></tr> <tr><td></td><td>D8004</td><td>2</td><td>JACK</td></tr> <tr><td></td><td>D05-04</td><td>1</td><td>JACK</td></tr> <tr><td></td><td>MS02-2</td><td>2</td><td>JACK</td></tr> <tr><td></td><td>MS02-3</td><td>2</td><td>JACK</td></tr> <tr><td></td><td>98A07003903001</td><td>2</td><td>FITTING</td></tr> <tr><td></td><td>RTA 075009 LH</td><td>1</td><td>FITTING</td></tr> <tr><td></td><td>RTA 075010 RH</td><td>1</td><td>FITTING</td></tr> </tbody> </table> Refer to Figure 2 , Sheet 3 . B) AIRCRAFT CONFIGURATION This jacking procedure is only applicable to the following configuration: - The aircraft without: - Fuel (except undrainable fuel) - Payload - Water (potable) - Baggage containers - Pallets - Emergency equipment - Catering - Documents and aircraft tool package			ITEM	PART NUMBER	QTY	DESCRIPTION		343A160KN2	1	LOAD CELL		343A400KN2	2	LOAD CELL		343A100KN2	4	LOAD CELL		343A63KN2	1	LOAD CELL		98A07003910200	1	ADAPTOR		98A07003906200	1	ADAPTOR		98A07004002000	2	ADAPTOR		98A07003909200	2	ADAPTOR		98A07003911200	2	ADAPTOR		98A07003907200	5	ADAPTOR		98A07003906200	5	ADAPTOR		D30-11	1	JACK		D8004	2	JACK		D05-04	1	JACK		MS02-2	2	JACK		MS02-3	2	JACK		98A07003903001	2	FITTING		RTA 075009 LH	1	FITTING		RTA 075010 RH	1	FITTING		
ITEM	PART NUMBER	QTY	DESCRIPTION																																																																																		
	343A160KN2	1	LOAD CELL																																																																																		
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Sıra No Item No	İş Tarifi Accomplishment Instructions			Uygulayan Perform By	Kontrol Eden Inspected By																				
	For the complete aircraft configuration: Refer to Figure 2 , Sheet 1 NOTE: If the aircraft configuration is not the same as the configuration above, contact Engineering. NOTE: If you should do other tasks on the aircraft at the same time, contact Engineering. C) JACKING PROCEDURE - Obey the general jacking instructions (Refer to SRM 51-51-00 Page block 001). - Lift the aircraft on jacks at points A, B1, B2 and C as shown in Figure 2, Sheet 2 (Refer to AMM 07-11-00 Page block 001). - Install the additional jacking equipment at FR17 and FR76 (Refer to Figure 2, Sheets 2 and Sheet 3). CAUTION: MAKE SURE THAT THE LOADS NEVER EXCEED THE MAXIMUM PERMITTED LOADS SPECIFIED IN FIGURE 2, SHEET 4. - Apply the loads shown in step 1, to the jacks at FR17, FR76 and FR85. Refer to Figure 2, Sheet 4. - Wait until the loads on the jacks become stable and make sure that they are within a tolerance of +/- 10 percent. - Do the same procedure to apply the loads shown in step 2, step 3 and step 4. NOTE: When the loads given in step 4 are stable and within a tolerance of +/- 10 percent, you can do the procedure given in the Para C. PROCEDURES 2. MODIFICATION. 2. MODIFICATION NOTE: If you find existing doubler(s) (e.g. skin repairs) with a distance less than 4 rivet pitches between the outer rivet-rows of the new doublers and the repair doublers, contact Engineering. Refer to Kit 532125D01R02 - Install the external doubler at FR55/58, between STGR13, LH and STGR13, RH. Refer to Figure 1 . - Temporarily put these items in position and mark their outer profile on the outer fuselage skin: <table border="0"> <tr> <td>b.</td> <td>2</td> <td>Shim</td> <td>A5397986820200</td> <td>Item 1</td> </tr> <tr> <td></td> <td>2</td> <td>Shim</td> <td>A5397986820000</td> <td>Item 2</td> </tr> <tr> <td></td> <td>2</td> <td>Strap</td> <td>A5397986720200</td> <td>Item 3</td> </tr> <tr> <td></td> <td>1</td> <td>Strap</td> <td>A5397986720000</td> <td>Item 4</td> </tr> </table> NOTE: Make sure that the edge-distance of the fasteners is in accordance with the SRM 51-40-50.			b.	2	Shim	A5397986820200	Item 1		2	Shim	A5397986820000	Item 2		2	Strap	A5397986720200	Item 3		1	Strap	A5397986720000	Item 4		
b.	2	Shim	A5397986820200	Item 1																					
	2	Shim	A5397986820000	Item 2																					
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	1	Strap	A5397986720000	Item 4																					
Uçak Tes. No A/C Reg. No		THY P/K THY Part Number		Bakım No Maint. Check No																					
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TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	6																																																																																				
Sıra No Item No	İş Tarifi Accomplishment Instructions			Uygulayan Perform. By	Kontrol Eden Inspected By																																																																																				
	NOTE: Make sure that the fastener spacing is the same as the existing pitch or in accordance with the SRM 51-40-50. c. Identify the fasteners that you must remove in the modification area with a wax pencil. d. Remove and discard the fasteners. e. Temporarily put in position on the outer fuselage skin: <table border="0"> <tr> <td>2</td> <td>Shim</td> <td>A5397986820200</td> <td>Item 1</td> </tr> <tr> <td>2</td> <td>Shim</td> <td>A5397986820000</td> <td>Item 2</td> </tr> <tr> <td>2</td> <td>Strap</td> <td>A5397986720200</td> <td>Item 3</td> </tr> <tr> <td>1</td> <td>Strap</td> <td>A5397986720000</td> <td>Item 4</td> </tr> </table> and transfer drill the existing and new fastener holes. f. Remove the shims, Item 1, the shims, Item 2, the straps, Item 3 and the strap, Item 4 and deburr them. g. Countersink the straps, Item 3, and the straps, Item 4. h. Fill the existing countersinks in the outer fuselage skin with countersink repair washers (Refer to SRM 51-71-00 Page block 201). i. Apply a thin layer of Sealant (Mat No. 09-013) to all the mating surfaces. j. Put in position on the outer fuselage skin: <table border="0"> <tr> <td>2</td> <td>Shim</td> <td>A5397986820200</td> <td>Item 1</td> </tr> <tr> <td>2</td> <td>Shim</td> <td>A5397986820000</td> <td>Item 2</td> </tr> <tr> <td>2</td> <td>Strap</td> <td>A5397986720200</td> <td>Item 3</td> </tr> <tr> <td>1</td> <td>Strap</td> <td>A5397986720000</td> <td>Item 4</td> </tr> </table> and wet install with: <table border="0"> <tr> <td>738</td> <td>Pin-Rivet</td> <td>DAN6-6-3</td> <td>Item 5</td> </tr> <tr> <td>40</td> <td>Pin-Rivet</td> <td>DAN6-6-3X</td> <td>Item 17</td> </tr> <tr> <td>496</td> <td>Pin-Rivet</td> <td>DAN6-6-5X</td> <td>Item 7</td> </tr> <tr> <td>218</td> <td>Pin-Rivet</td> <td>DAN6-6-7X</td> <td>Item 9</td> </tr> <tr> <td>174</td> <td>Pin-Rivet</td> <td>DAN6-6-6X</td> <td>Item 10</td> </tr> <tr> <td>74</td> <td>Pin-Rivet</td> <td>DAN6-6-8X</td> <td>Item 11</td> </tr> <tr> <td>36</td> <td>Pin-Rivet</td> <td>DAN6-6-7</td> <td>Item 12</td> </tr> <tr> <td>24</td> <td>Pin-Rivet</td> <td>DAN6-6-4X</td> <td>Item 13</td> </tr> <tr> <td>16</td> <td>Pin-Rivet</td> <td>DAN6-6-10X</td> <td>Item 14</td> </tr> <tr> <td>14</td> <td>Pin-Rivet</td> <td>DAN6-6-9X</td> <td>Item 15</td> </tr> <tr> <td>10</td> <td>Pin-Rivet</td> <td>DAN6-6-8</td> <td>Item 16</td> </tr> <tr> <td>1056</td> <td>Collar</td> <td>DAN11-6X</td> <td>Item 8</td> </tr> <tr> <td>784</td> <td>Collar</td> <td>NSA5075-6</td> <td>Item 6</td> </tr> </table> NOTE: For fastener identification, refer to Figure 1, Sheet 10 . k. Remove any excess sealant with a Lint-Free Cotton Cloth (Mat No. 19-003) and Solvent General Purpose (Mat No. 11-026) .			2	Shim	A5397986820200	Item 1	2	Shim	A5397986820000	Item 2	2	Strap	A5397986720200	Item 3	1	Strap	A5397986720000	Item 4	2	Shim	A5397986820200	Item 1	2	Shim	A5397986820000	Item 2	2	Strap	A5397986720200	Item 3	1	Strap	A5397986720000	Item 4	738	Pin-Rivet	DAN6-6-3	Item 5	40	Pin-Rivet	DAN6-6-3X	Item 17	496	Pin-Rivet	DAN6-6-5X	Item 7	218	Pin-Rivet	DAN6-6-7X	Item 9	174	Pin-Rivet	DAN6-6-6X	Item 10	74	Pin-Rivet	DAN6-6-8X	Item 11	36	Pin-Rivet	DAN6-6-7	Item 12	24	Pin-Rivet	DAN6-6-4X	Item 13	16	Pin-Rivet	DAN6-6-10X	Item 14	14	Pin-Rivet	DAN6-6-9X	Item 15	10	Pin-Rivet	DAN6-6-8	Item 16	1056	Collar	DAN11-6X	Item 8	784	Collar	NSA5075-6	Item 6		
2	Shim	A5397986820200	Item 1																																																																																						
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24	Pin-Rivet	DAN6-6-4X	Item 13																																																																																						
16	Pin-Rivet	DAN6-6-10X	Item 14																																																																																						
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Uçak Tes. No A/C Reg. No		THY P/K THY Part Number		Bakım No Maint. Check No																																																																																					
İş Emri No Work Order No		THY S/N THY Serial Number		Bakım Tarihi Maint. Date																																																																																					

Issue Date : 23.05.2006

Form No: MPM 1.8A/01B

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	9/32	



TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	7
Sıra No Item No	İş Tarifi Accomplishment Instructions			Uygulayan Perform. By	Kontrol Eden Inspected By
	TEST a) Do a leak check of the removed air conditioning duct(s). NOTE: Do this test after the installation of the air conditioning duct(s). Refer to paragraph D. 1) Make sure that the aircraft electrical circuits are energized. (Refer to AMM 24-41-00 Page block 301). 2) Make sure that the ventilation for the electronics rack is correct. 3) On the panel 438VU, push in the CABIN FANS pushbutton switch. 4) Make sure that there are no leaks at the connections of the removed airconditioning duct(s). 5) On the panel 438VU, release out the CABIN FANS pushbutton switch.				
D	CLOSE UP For passenger Aircrafts a) Make sure that the work area is clean and clear of tools and other items of equipment. b) Install the removed heat and sound insulation. (Refer to AMM 25-22-11 Page block 401). c) Install the removed air conditioning duct(s). d) Instal the removed ceiling panels (Refer to AMM 25-23-11 Page block 401). e) Install the removed sidewall OHSC (Refer to AMM 25-24-11 Page block 401). f) Install the removed center OHSC (Refer to AMM 25-24-21 Page block 401). g) Lower the aircraft to the ground (Refer to E. 3. De-jacking). f) Fill the result Information Form. 3. De-jacking a) Obey the general de-jacking instructions (Refer to SRM 51-51-00 Page block 001). b) Gradually decrease the loads one step after the other, from step 4 to zero on the jacks at FR17, FR76 and FR85. Refer to Figure 2, Sheet 4. c) Remove the jacking equipment at FR17 and FR76. d) Lower the aircraft and remove jacking equipment at points A, B1, B2 and C as shown in Figure 2, Sheet 2 (Refer to AMM 07-11-00 Page block 001).				
Uçak Tes. No A/C Reg. No		THY P/K THY Part Number		Bakım No Maint. Check No	
İş Emri No Work Order No		THY S/N THY Serial Number		Bakım Tarihi Maint. Date	

Issue Date : 23.05.2006

Form No: MPM 1.8A/01B

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	10/32	



MANDATORY

CMT (II)

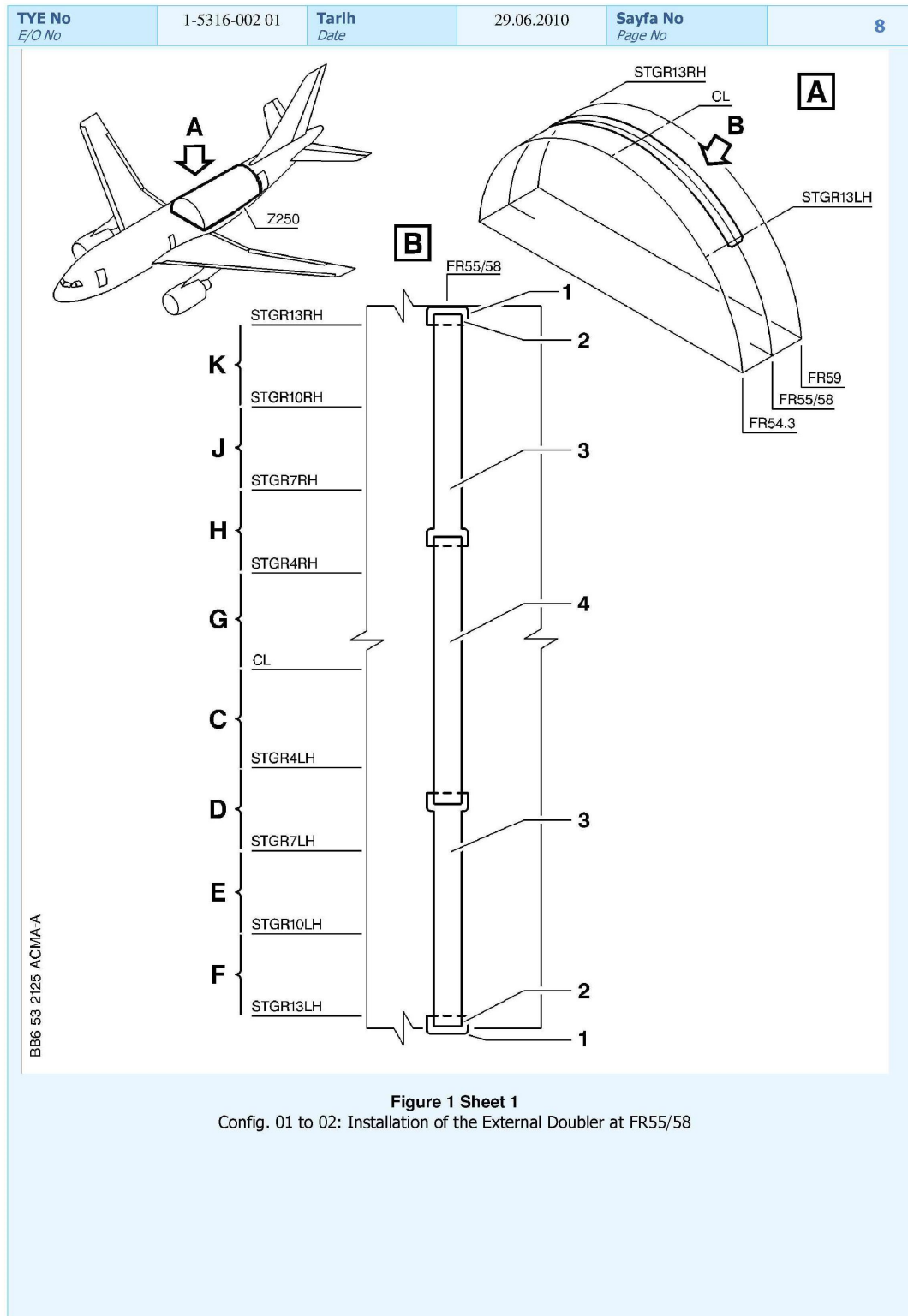


TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58



Issue Date : 23.05.2006

Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	11/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20





MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	9
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C

NOTE: FOR FASTENER SYMBOLS, REFER TO SHEET 10

Figure 1 Sheet 2
Config. 01 to 02: Installation of the External Doubler at FR55/58

Form No: MPM 1.8A/01C

Issue Date : 23.05.2006

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	12/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20





MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	10
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D

NOTE: FOR FASTENER SYMBOLS, REFER TO SHEET 10
FOR DIMENSION TABLE, REFER TO SHEET 10

Figure 1 Sheet 3
Config. 01 to 02: Installation of the External Doubler at FR55/58

Issue Date : 23.05.2006 Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	13/32	

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Form No : MPM 1.8D/01 R20





MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	11
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E

NOTE: FOR FASTENER SYMBOLS, REFER TO SHEET 10

Figure 1 Sheet 4
Config. 01 to 02: Installation of the External Doubler at FR55/58

Issue Date : 23.05.2006

Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	14/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20





MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

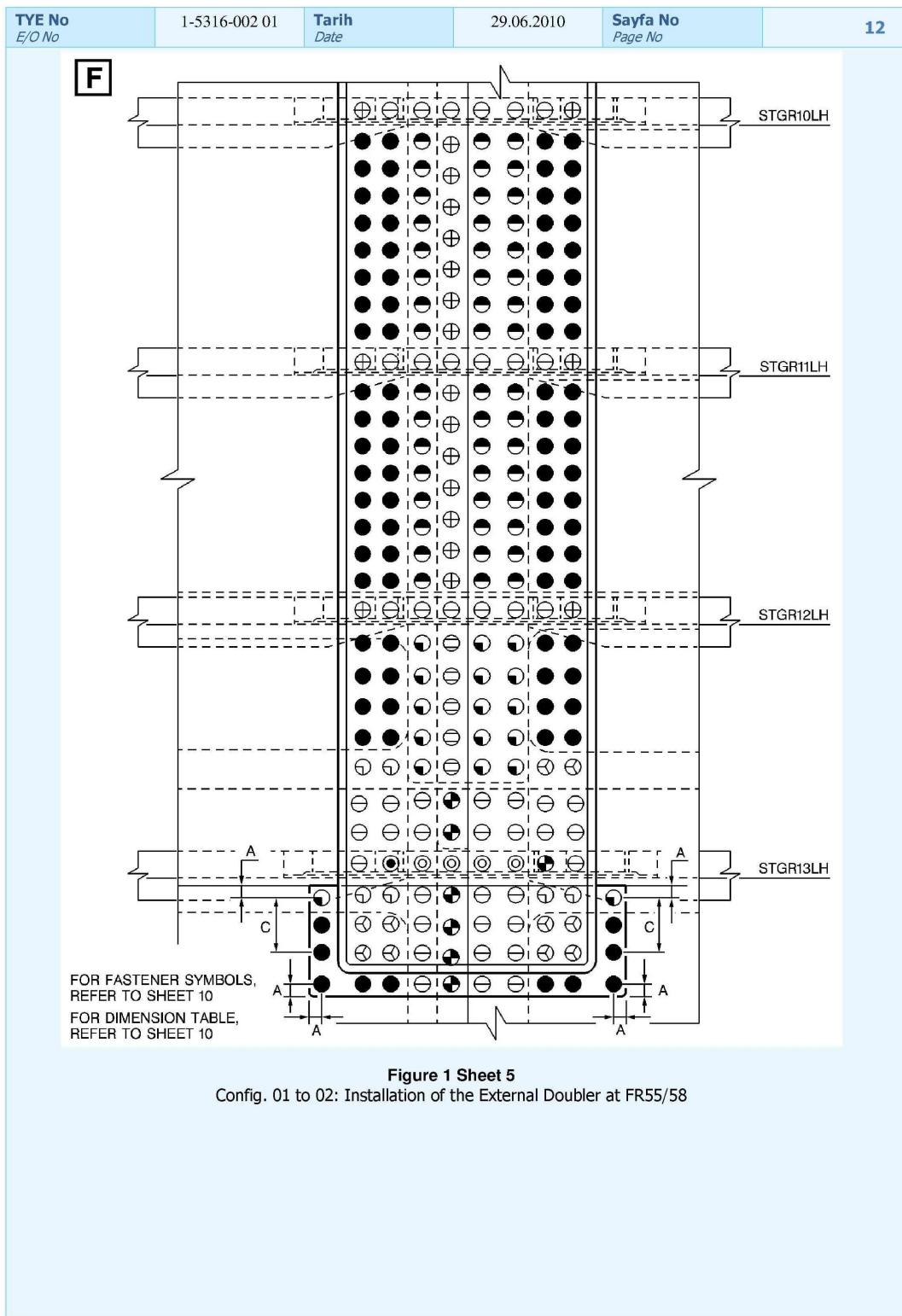


Figure 1 Sheet 5
Config. 01 to 02: Installation of the External Doubler at FR55/58

Issue Date : 23.05.2006

Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	15/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20





TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	13
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NOTE: FOR FASTENER SYMBOLS, REFER TO SHEET 10

Figure 1 Sheet 6
 Config. 01 to 02: Installation of the External Doubler at FR55/58

Form No: MPM 1.8A/01C

Issue Date : 23.05.2006

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	16/32	

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Form No : MPM 1.8D/01 R20



MANDATORY

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TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	14
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H

NOTE: FOR FASTENER SYMBOLS, REFER TO SHEET 10
FOR DIMENSION TABLE, REFER TO SHEET 10

Figure 1 Sheet 7
Config. 01 to 02: Installation of the External Doubler at FR55/58

Issue Date : 23.05.2006 Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	17/32	

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Form No : MPM 1.8D/01 R20





MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	15
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J

NOTE: FOR FASTENER SYMBOLS, REFER TO SHEET 10

Figure 1 Sheet 8
Config. 01 to 02: Installation of the External Doubler at FR55/58

Issue Date : 23.05.2006

Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	18/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20





MANDATORY

CMT (II)

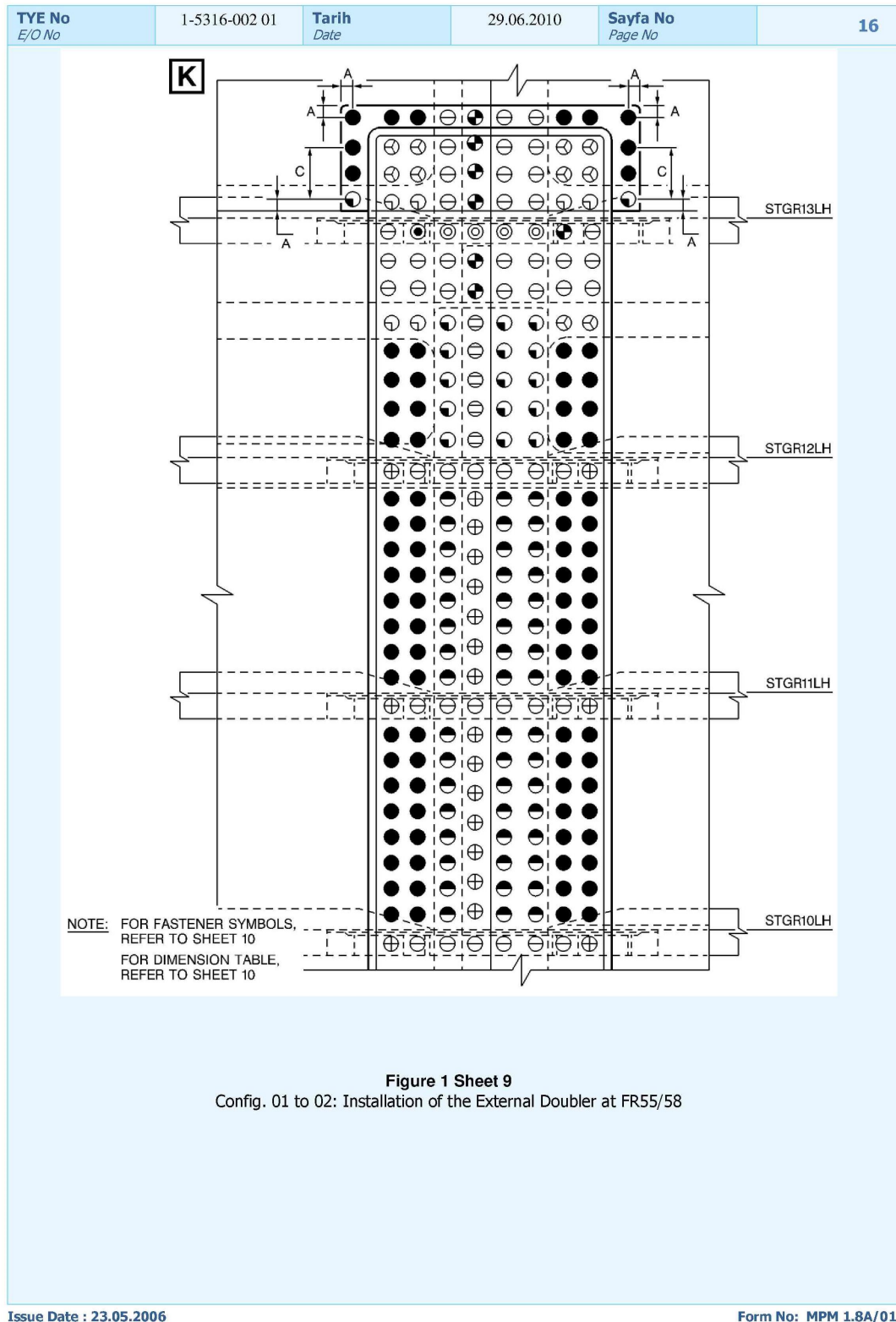


TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58



ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	19/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20





TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	17		
HOLE SYMBOL	ITEM	P/N	DESCRIPTION	KIT	HOLE DIAMETER		REMARKS
					MIN	MAX	
●	5	DAN6 6 3	PIN	D01	4.78mm (0.188in.)	4.82mm (0.190in.)	NOMINAL SIZE TRANSITION FIT
	6	NSA5075 6	COLLAR				
◐	7	DAN6 6 5X	PIN	D01	5.11mm (0.201in.)	5.15mm (0.203in.)	1ST OVERSIZE TRANSITION FIT
	8	DAN11 6X	COLLAR				
⊖	9	DAN6 6 7X	PIN	D01	5.11mm (0.201in.)	5.15mm (0.203in.)	1ST OVERSIZE TRANSITION FIT
	8	DAN11 6X	COLLAR				
⊕	10	DAN6 6 6X	PIN	D01	5.11mm (0.201in.)	5.15mm (0.203in.)	1ST OVERSIZE TRANSITION FIT
	8	DAN11 6X	COLLAR				
⊗	11	DAN6 6 8X	PIN	D01	5.11mm (0.201in.)	5.15mm (0.203in.)	1ST OVERSIZE TRANSITION FIT
	8	DAN11 6X	COLLAR				
◑	12	DAN6 6 7	PIN	D01	4.78mm (0.188in.)	4.82mm (0.190in.)	NOMINAL SIZE TRANSITION FIT
	6	NSA5075 6	COLLAR				
⦿	13	DAN6 6 4X	PIN	D01	5.11mm (0.201in.)	5.15mm (0.203in.)	1ST OVERSIZE TRANSITION FIT
	8	DAN11 6X	COLLAR				
◎	14	DAN6 6 10X	PIN	D01	5.11mm (0.201in.)	5.15mm (0.203in.)	1ST OVERSIZE TRANSITION FIT
	8	DAN11 6X	COLLAR				
⦿	15	DAN6 6 9X	PIN	D01	5.11mm (0.201in.)	5.15mm (0.203in.)	1ST OVERSIZE TRANSITION FIT
	8	DAN11 6X	COLLAR				
⊖	16	DAN6 6 8	PIN	D01	4.78mm (0.188in.)	4.82mm (0.190in.)	NOMINAL SIZE TRANSITION FIT
	6	NSA5075 6	COLLAR				
⦿	17	DAN6 6 3X	PIN	D01	5.11mm (0.201in.)	5.15mm (0.203in.)	1ST OVERSIZE TRANSITION FIT
	8	DAN11 6X	COLLAR				

DIMENSIONS		
	mm	(in.)
A	10.0	0.394
B	16.0	0.629

DIMENSIONS		
	mm	(in.)
C	2x20.0	2x0.787
D	3x20.0	3x0.787

Figure 1 Sheet 10
 Config. 01 to 02: Installation of the External Doubler at FR55/58

Issue Date : 23.05.2006

Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	20/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20



TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	18
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A/C CONFIGURATION FOR JACKING

FUSELAGE		INSTALLED	REMOVED	LH WING		INSTALLED	REMOVED	RH WING		INSTALLED	REMOVED
EQUIPMENT/ FURNISHINGS				BELLY FAIRING				BELLY FAIRING			
FUSELAGE SECTION 11		X		LH SLAT		X		RH SLAT		X	
FUSELAGE SECTION 12		X		LH FLAP		X		RH FLAP		X	
FUSELAGE SECTION 13		X		LH FLAP TRACKS		X		RH FLAP TRACKS		X	
FUSELAGE SECTION 14		X		SPOILER		X		SPOILER		X	
FUSELAGE SECTION 15		X									
FUSELAGE SECTION 16		X									
FUSELAGE SECTION 17		X									
FUSELAGE SECTION 18		X									
FUSELAGE SECTION 19		X									

MOVABLE ELEMENTS		INSTALLED	REMOVED	OTHERS		INSTALLED	REMOVED
HORIZONTAL STABILIZER		X		LANDING GEARS			
VERTICAL STABILIZER		X		NOSE LANDING GEAR		X	
FWD LH PAX DOOR		X		LH MAIN LANDING GEAR		X	
FWD RH PAX DOOR		X		RH MAIN LANDING GEAR		X	
LH EMERGENCY EXIT		X		PYLONS			
RH EMERGENCY EXIT		X		LH PYLON		X	
AFT LH PAX DOOR		X		RH PYLON		X	
AFT RH PAX DOOR		X		ENGINES			
FWD CARGO DOOR		X		LH ENGINE		X	
AFT CARGO DOOR		X		RH ENGINE		X	
APU		X		THRUST REVERSERS			
				LH ENGINE		X	
				RH ENGINE		X	

FUEL		
LH WING TANKS		0
RH WING TANKS		0
TRIM TANK (WHERE INSTALLED)		0
EXCEPT UNDRAINABLE FUEL		

Figure 2 Sheet 1

Jacking of the Aircraft for the Installation of the External Doubler at FR55/58

Issue Date : 23.05.2006

Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	21/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20



MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	19
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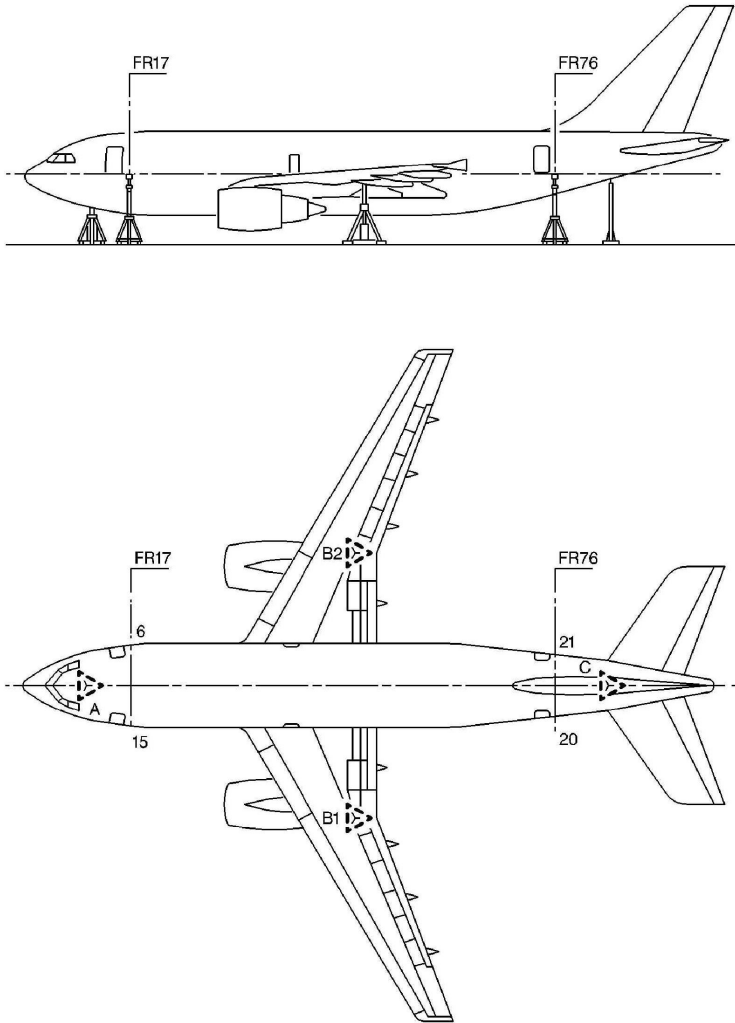


Figure 2 Sheet 2
Jacking of the Aircraft for the Installation of the External Doubler at FR55/58

Issue Date : 23.05.2006

Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	22/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20





MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	20
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JACKING OR SUPPORT POINTS	POINT	CRADLE OR FITTING PART NUMBER	JACK PART NUMBER	QTY	LOAD CELL ADAPTOR	QTY	LOAD CELL PART NUMBER	QTY
FRONT JACK	A	N/A	D30-11	1	98A07003910200	1	343A160KN2	1
LH MAIN JACK	B1	N/A	D8004	1	98A07004002000	1	343A400KN2	1
RH MAIN JACK	B2	N/A	D8004	1	98A07004002000	1	343A400KN2	1
SAFETY (FR85)	C	N/A	D05-04	1	98A07003907200	1	343A63KN2	1
JACK FR17 LH	15	98A07003903001	MS02-2	1	98A07003906200	1	343A100KN2	1
JACK FR17 RH	6	98A07003903001	MS02-2	1	98A07003907200	1	343A100KN2	1
JACK FR76 LH	20	RTA 075009 LH	MS02-3	1	98A07003907200	1	343A100KN2	1
JACK FR76 RH	21	RTA 075010 RH	MS02-3	1	98A07003907200	1	343A100KN2	1

N/A: NOT APPLICABLE

Figure 2 Sheet 3

Jacking of the Aircraft for the Installation of the External Doubler at FR55/58

Issue Date : 23.05.2006

Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	23/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechinc.com

Form No : MPM 1.8D/01 R20





TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	21
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JACKING POINTS	MAX. LOAD (daN)	APPLIED LOADS (daN) +/- 10%				% OF TOTAL LOADS	ACTUAL LOADS
		STEP 1	STEP 2	STEP 3	STEP 4		
FR17 LH	8250	1000	2000	3000	4000	5	
FR17 RH	8250	1000	2000	3000	4000	5	
FR76 LH	5200	750	1500	2250	3000	4	
FR76 RH	5200	750	1500	2250	3000	4	
FR85 (SAFETY JACK)	4500	1000	2000	3000	4000	5	

NOTE 1: TO INSTALL KIT 532125D01 (REFER TO ACCOMPLISHMENT INSTRUCTION PARAGRAPH B.), MAKE SURE THAT THE ACTUAL LOADS ARE WITHIN A TOLERANCE OF +/- 10% OF THE VALUES GIVEN IN STEP 4.

NOTE 2: APPLIED LOADS ARE THE TOTAL LOAD AS PER STATION (FOR A STATION WITH 2 JACKS, THE LOAD FOR EACH JACK IS THE APPLIED LOAD DIVIDED BY TWO).

SUPPORT POINTS	MAX. LOAD (daN)	REMAINING LOADS (daN)			
FRONT JACK	12000	FOR INFORMATION		10500	13
LH MAIN JACK	59600	FOR INFORMATION		25500	32
RH MAIN JACK	59600	FOR INFORMATION		25500	32

CORRESPONDING WEIGHT OF THE AIRCRAFT 79500 Kg (INCLUDING ANY BALLAST)

Figure 2 Sheet 4

Jacking of the Aircraft for the Installation of the External Doubler at FR55/58

Issue Date : 23.05.2006

Form No: MPM 1.8A/01C

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	24/32	

MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

THY TEKNİK / TURKISH TECHNIC

Approval References : EASA.145.0276 SHY-145/JAR-145; TR-00001; FAA: TQKY144F

HAZIRLANAN VE PARÇA DEĞİŞİKLİĞİ FORMU
MATERIAL AND PART PROCUREMENT INSTRUCTIONS

TYE No
E/O No

1-5316-002 01

Tarih
Date

29.06.2010

Sayfa No
Page No

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Sıra No Item No	Eski Parça / Old Part		Parçanın Adı Nomenclature	Birim Mik. Unit Qty.	Değ. Kodu Change Code	Dur. Kodu Status Code	Yeni Parça / New Part		Birim Mik. Unit Qty.	Top. Sipariş Total Order	Depo Yed. Spare Qty.	Notlar Notes
	THY P/K	MFR P/N					THY P/K	MFR P/N				
A			MOD KIT					532125D01R02	1	2		Note1,3
			KIT INCLUDING					532125D01R02				
A.1			Shim					A5397986820200	2			
A.2			Shim					A5397986820000	2			
A.3			Strap					A5397986720200	2			
A.4			Strap					A5397986720000	1			
A.5			Pin					DAN6-6-3	738			
A.6			Collar					DAN6-6-3	784			
A.7			Pin					DAN6-6-5X	496			
A.8			Collar					DAN11-6X	1056			
A.9			Pin					DAN6-6-7X	218			
A.10			Pin					DAN6-6-6X	174			
A.11			Pin					DAN6-6-8X	74			
A.12			Pin					DAN6-6-7	36			
A.13			Pin					DAN6-6-4X	24			
A.14			Pin					DAN6-6-10X	16			
A.15			Pin					DAN6-6-9X	14			
A.16			Pin					DAN6-6-8	10			
A.17			Pin					DAN6-6-3X	40			

Değişim Kodları/ Change Codes				Durum Kodları / Status Codes			
A	Tamamen I/C - Fully I/C	E	İlave edildi/ Added	N	Depoya faal iade/ To be returned to Stock as Serviceable		
B	Tek yönlü I/C - One-way I/C	F	Miktar Değişti/ Qty. Changed	O	KKM' ne gayrifaal iade/ To be returned to Component Control Center as Unserviceable		
C	I/C değil/ Not I/C	G	Diğer/ Other	R	Satılacak / To be Sold	S	Scrap edilecek/ To be Scrapped
D	İptal edildi/ Deleted			G	Diğer/ Other		

Issue Date : 28.06.2007

Form No: MPM 1.8A/01D R2

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	25/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechinc.com

Form No : MPM 1.8D/01 R20





TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

THY TEKNİK / TURKISH TECHNIC

Approval References : EASA.145.0276 SHY-145/JAR-145: TR-00001; FAA: TQK144F

 MALZEME VE PARÇA DEĞİŞİKLİĞİ FORMU
 MATERIAL AND PART PROCUREMENT INSTRUCTIONS

TYE No E/O No		1-5316-002 01		29.06.2010		Sayfa No Page No		23	
Tarih Date		Yeni Parça / New Part		THY P/K		MFR P/N		Top. Sipariş Total Order	
Dur. Kodu Status Code		Depo Yed. Spare Qty.		Notlar Notes		Birim Mik. Unit Qty.		Birim Mik. Unit Qty.	
Parçanın Adı Nomenclature		Eski Parça / Old Part		THY P/K		MFR P/N		Birim Mik. Unit Qty.	
CONSUMABLE MATERIALS									
B.1	Sealant			C991686		CML 09-013		AR	Note 1,2
B.2	Solvent General Purpose			C992186		CML 11-026		AR	Note 1,2
B.3	Anti-Corrosion Primer					CML 16-001		AR	Note 1,2
B.4	Polyurethane Top Coat Grey (For Internal Applic.)					CML 16-002		AR	Note 1,2
B.5	External Top Coat			C990096		CML 16-018		AR	Note 1,2
B.6	Wash Primer					CML 16-020		AR	Note 1,2
B.7	Lint-Free Cotton Cloth					CML 19-003		AR	Note 1,2
	JACKING TOOLS								
C	JACK FR17 LH / RH								
C.1	CRADLE OR FITTING			T132506		98A07003903001		2	
C.2	JACK			T132507		MS02-2		2	
C.3	LOAD CELL ADAPTOR			T132508		98A07003907200		2	
C.4	LOAD CELL ADAPTOR			T132509		98A07003906200		2	
C.5	LOAD CELL			T132511		343A100KN2		2	

Durum Kodları / Status Codes	
N	Depoya faal iade/ To be returned to Stock as Serviceable
O	KKM' ne gayrifaal iade/ To be returned to Component Control Center as Unserviceable
R	Satılacak / To be Sold
G	Scrap edilecek/ To be Scrapped
	Diger/ Other

Değişim Kodları / Change Codes	
A	Tamamen I/C - Fully I/C
B	Tek yönlü I/C - One-way I/C
C	I/C değil / Not I/C
D	İptal edildi / Deleted
E	İlave edildi / Added
F	Miktarı Değişt / Qty. Changed
G	Diger / Other

Form No: MPM 1.8A/01DR2

Issue Date : 28.06.2007

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	26/32	

MANDATORY

CMT (II)



TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

THY TEKNİK / TURKISH TECHNIC

Approval References : EASA.145.0276 SHY-145/JAR-145: TR-00001; FAA: TQK144F


 MALZEME VE PARÇA DEĞİŞİKLİĞİ FORMU
 MATERIAL AND PART PROCUREMENT INSTRUCTIONS

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	24
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Sıra No Item No	Eski Parça / Old Part		Parçanın Adı Nomenclature	Birim Mik. Unit Qty.	Deg. Kodu Change Code	Dur. Kodu Status Code	Yeni Parça / New Part		Birim Mik. Unit Qty.	Top. Sipariş Total Order	Depo Yed. Spare Qty.	Notlar Notes
	THY P/K	MFR P/N					THY P/K	MFR P/N				
D			JACK FR76 LH / RH				T153570	RTA 075009 LH	1			
			CRADLE OR FITTING				T153572	RTA 075010 RH	1			
			CRADLE OR FITTING				T153569	MS02-3	2			
			JACK				T132508	98A07003907200	2			
			LOAD CELL ADAPTOR				T132509	98A07003906200	2			
			LOAD CELL ADAPTOR				T132511	343A100KN2	2			
			LOAD CELL									
E			MAIN JACK LH / RH									
E1			CRADLE OR FITTING					N/A				
E2			JACK					D8004	2			
E3			LOAD CELL ADAPTOR				T107001	98A07004002000	2			
E4			LOAD CELL ADAPTOR				T153568	98A07003911200	2			
E5			LOAD CELL ADAPTOR				T153566	98A07003909200	2			
E6			LOAD CELL				T153565	343A400KN2	2			

Değişim Kodları / Change Codes			
A	Tamamen I/C - Fully I/C	E	İlave edildi / Added
B	Tek yönlü I/C - One-way I/C	F	Miktarı Değişt / Qty. Changed
C	I/C değil / Not I/C	G	Diğer / Other
D	İptal edildi / Deleted		

Issue Date : 28.06.2007

Durum Kodları / Status Codes			
N	Depoya faal iade / To be returned to Stock as Serviceable		
O	KKM' ne gayrifaal iade / To be returned to Component Control Center as Unserviceable		
R	Satılacak / To be Sold	S	Scrap edilecek / To be Scrapped
G	Diğer / Other		

Form No: MPM 1.8A/01DR2

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	27/32	

Prepared by Turkish Airlines Technic Inc. www.turkishairlines.com

Form No : MPM 1.8D/01 R20



TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

THY TEKNİK / TURKISH TECHNIC

Approval References : EASA.145.0276 SHY-145/JAR-145: TR-00001; FAA: TQK144F

 MALZEME VE PARÇA DEĞİŞİKLİĞİ FORMU
 MATERIAL AND PART PROCUREMENT INSTRUCTIONS

TYE No E/O No	1-5316-002 01	Tarih Date	29.06.2010	Sayfa No Page No	25
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Sıra No Item No	Eski Parça / Old Part		Parçanın Adı Nomenclature	Birim Mik. Unit Qty.	Deg. Kodu Change Code	Dur. Kodu Status Code	Yeni Parça / New Part		Birim Mik. Unit Qty.	Top. Sipariş Total Order	Depo Yed. Spare Qty.	Notlar Notes
	THY P/K	MFR P/N					THY P/K	MFR P/N				
F			FRONT JACK									
F.1			CRADLE OR FITTING					N/A				
F.2			JACK					D30-11	1			
F.3			LOAD CELL ADAPTOR				T153567	98A07003910200	1			
F.4			LOAD CELL ADAPTOR				T132509	98A07003906200	1			
F.5			LOAD CELL				T953535	343A160KN2	1			
G			SAFETY JACK (FR85)									
G.1			CRADLE OR FITTING					N/A				
G.2			JACK					D05-04	1			
G.3			LOAD CELL ADAPTOR				T132508	98A07003907200	1			
G.4			LOAD CELL ADAPTOR				T132509	98A07003906200	1			
G.5			LOAD CELL				T132510	343A63KN2	1			
G.6												

Notlar / Notes												
Note 1: AR: As Required. Note 2: Check availability of all materials before assigning this to any A/C. Note 3: For purchase orders placed before 31 Dec. 2018, Airbus will provide Kits with 35 % reduction.												

Değişim Kodları / Change Codes				Durum Kodları / Status Codes			
A	Tamamen I/C - Fully I/C	E	İlave edildi / Added	N	Depoya faal iade / To be returned to Stock as Serviceable		
B	Tek yönlü I/C - One-way I/C	F	Miktarı Değişt / Qty. Changed	O	KKM' ne gayrifaal iade / To be returned to Component Control Center as Unserviceable		
C	I/C değil / Not I/C	G	Diğer / Other	R	Satılacak / To be Sold	S	Scrap edilecek / To be Scrapped
D	İptal edildi / Deleted			G	Diğer / Other		

Form No: MPM 1.8A/01D R2

Issue Date : 28.06.2007

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	28/32	



TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58



THY TEKNİK / TURKISH TECHNIC

Approval References : EASA.145.0276 SHY-145/JAR-145: TR-00001; FAA: TQKY144F

SONUÇ BİLDİRİM FORMU		RESULT INFORMATION FORM	
Teknik Yayın No Tech. Document No	1-5316-002 01	Tarih Date	29.06.2010
Sayfa No Page No		Sayfa No Page No	26
AD-CN No Alert-Mandatory SB	EASA 2007-0111 00, AD 2007-25-09 00, BA310 53-2125 01 EASA 2007-0111 Rev:00 AD 2007-25-09 Rev:00		
SBF'yi Dolduracak Bölüm / RIF is filled by : Structural Repair Shop			
SBF Dağıtım Listesi / RIF Distribution List			
Bilgi / Info	YAPISAL MUHENDISLIK , WARRANTY GARANTI GRUBU , UPK MD.LUGU ,		
Gereği / Action Required			
SBF mutlaka doldurulmalıdır. Bulgular için bölge, istasyon, ve yön belirtilmelidir. <i>RIF should not be left blank even if no findings exist. Zone, station, and direction/side should be indicated</i>			
SONUC VE BULGULAR/ RESULTS & FINDINGS :			

	Adı Soyadı / Name & Surname	Tarih / Date	İmza / Signature
Uygulayan/ Performed By			
Onaylayan/ Approved By			

DİKKAT : BU KISIM YAYIN "AD/ALERT-MANDATORY SB REFERANSLI" İŞE İŞİ YAPAN KİŞİ TARAFINDAN DOLDURULACAKTIR.
 WARNING : For AD/Alert-Mandatory SB related issues, this section should be filled by technician accomplishing this order.

	Uçak / Aircraft	Komponent / Component
Uygulama Saati Accomp. Time	Toplam Uçuş Saati Total F/H	Toplam İniş Sayısı Total F/C

Uçak Tes. No Aircraft Reg. No	THY P/K THY Part Code	Bakım No Maint. Check No
İş Emri No Work Order No	THY S/N THY Serial Number	Bakım Tarihi Maint. Date

Issue Date : 18.12.2009

Form No: MPM 1.8A/01E R4

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	29/32	



TASK CARD

 Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58**THY TEKNİK / TURKISH TECHNIC**
 Approval References: EASA.145.0276
 SHY-145/JAR-145: TR-00001; FAA: TQKY144F

TEKNİK YAYIN DAĞITIM FORMU
 TECHNICAL DOCUMENT DISTRIBUTION FORM

☒ TYE/EO	☐ MIF/ERF	☐ TBF/TIF	☐ KDF/CCC	☐ ÇP/ W/S	☐ Diğer/ Other	_____
Yayın No Document No	1-5316-002 01	Kayıt No Administration No		Tarih Date		
Bölmeler Departments	Adet Qty.	Tarih Date	Adı Soyadı Name & Surname	İmza Signature		
☒ Uçak Bakım Md.lüğü A/C Maintenance Mgmt.	1					
☐ Hat Bakım Dış İstasyonlar Md.lüğü Line Maintenance Ext.Stations Mgmt.	2					
☐ Tek. Doküman Yön. Md.lüğü (Sadece TBF) Technical Documents Mgmt. (Only TIF)	1					
☐ Hat Bakım Md.lüğü Line Maintenance Mgmt.	1					
☐ Uçak Bakım ERA Md.lüğü A/C Maintenance ERI Mngmt.	1					
☒ Yapısal Bakım ve Tahratsız Mua. Md.lüğü Structural Maintenance & NDT Mgmt.	1					
☒ Kabin içi Bakım Müdürlüğü Cabin Interior Maintenance Mgmt.	1					
☒ Bakım Planlama ve Kontrol Müdürlüğü Maintenance Planning and Control Mgmt.	2					
☐ Motor Revizyon Atelyeleri Md.lüğü Engine Overhaul W/Shops Mgmt.	1					
☐ H/M Komponent Atelyeleri Md.lüğü H/M Component W/Shops Mgmt.	1					
☐ Aviyonik Komponent Atelyeleri Md.lüğü Avionics Component W/Shops Mgmt.	1					
☒ Üretim Planlama ve Kontrol Md.lüğü Production Planning & Control Mngmt.	2					
☐ Teknik Pazarlama ve Satış Md.lüğü Technical Marketing & Sales Mgmt.	1					
☐ Depolar ve Stok Yön. Md.lüğü Stores & Inventory Mgmt.	1					
☐ İmalat Tamir Atl. Md.lüğü Production & Repair W/Shops Mgmt.	1					
☐ Dış Satınalma Md.lüğü Int'l Purchasing Mgmt.	1					
☐ VIP ve Hafif Uçaklar Md.lüğü VIP & Light A/C Mgmt.	1					
☐ Kalite Md.lüğü Quality Mgmt.	1					
☐ Eğitim Müdürlüğü Training Mgmt.	1					
☒ THY A.O. Teknik Bşk.lığı Turkish Airlines Inc., Tech. Directorate	1					
☐ THY A.O. Teknik Eğitim Md.lüğü Turkish Airlines Inc. Technical Training Mgmt.	1					
☐ THY A.O. Uçuş İşletme Bşk.lığı Turkish Airlines Inc. Flight Operations Directorate	1					
☐ THY A.O. Plan Performans Md.lüğü Turkish Airlines Inc. Plan Performance Mgmt.	1					
☐ THY A.O. Uçuş Eğitim Bşk.lığı Turkish Airlines Inc. Flight Training & Standards Dir.	1					
☐ THY A.O. Kabin Hizmetleri Bşk.lığı Turkish Airlines Inc. Cabin Services Directorate	1					

Issue Date: 05.09.2008

Form No: MPM 1.8A/04 R5

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	30/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20



MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

FORMS TO BE FILLED OUT:

- Fill out the Result Information Page (RIP) even if there is no finding. Should not be left blank.

TASK CARD FOLLOW UP:

Is there any finding related to this task card?: ☐ YES ☐ NO

If Yes, NRWI No:

Is there any Transferred Work Step (TWS) related to this task card?: ☐ YES ☐ NO

If Yes, Transferred Task Card /NRWI No:

Before you leave the aircraft, make sure that you take back all the materials left and tools you take with you while applying this task card,
FOD detected? ☐ YES ☐ NO

If Yes, specify FODs:

----- END OF THE TASK -----

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	31/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20





MANDATORY

CMT (II)



TASK CARD

Airline Card No.
A310-5316-002

Task Card Description

FUSELAGE - GENERAL - REINFORCE BUTT JOINT AT FR55/58

TASK CARD RESULT INFORMATION PAGE (RIP)

1 Confirm (select YES) if there is a finding related to nature of this EO/Task Card. 0

☐ Yes ☐ No

☒ Mandatory

CAUTION: If there is any finding, write here NRW no:

NAME, SURNAME:

Sign:

Date:

STAMP:

ACN	MSN	IPC	Work Order Number	Rev No.	Taskcard Rev Date	Page No	Seq. No
YA-CAV	0497	025	2636063	01	13.12.2014	32/32	

Prepared by Turkish Airlines Technic Inc. www.turkishtechnic.com

Form No : MPM 1.8D/01 R20

